
An Introduction to the Physics and Technology of e⁺e⁻ Linear Colliders

Lecture 9: a) Beam Based Alignment

Nick Walker (DESY)

USPAS, Santa Barbara, 16th-27th June, 2003

Emittance tuning in the LET

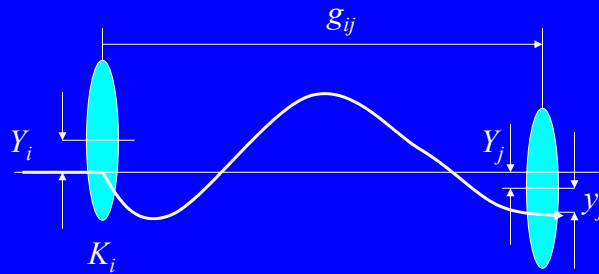
- LET = Low Emittance Transport
 - Bunch compressor (DR→Main Linac)
 - Main Linac
 - Beam Delivery System (BDS), inc. FFS
- DR produces tiny vertical emittances ($\gamma\varepsilon_y \sim 20\text{nm}$)
- LET must preserve this emittance!
 - strong wakefields (structure misalignment)
 - dispersion effects (quadrupole misalignment)
- Tolerances too tight to be achieved by surveyor during installation

⇒ Need beam-based alignment

mma!

Basics (linear optics)

thin-lens quad approximation: $\Delta y' = -KY$



$$g_{ij} = \left. \frac{\partial y_i}{\partial y_j'} \right|_{y_j'=0} = R_{34}(i, j)$$

$$y_j = \left(-\sum_{i=1}^j g_{ij} K_i Y_i \right) - Y_j$$

linear system: just superimpose oscillations caused by quad kicks.

Introduce matrix notation

Original Equation

$$y_j = \left(-\sum_{i=1}^j g_{ij} K_i Y_i \right) - Y_j$$

Defining Response Matrix \mathbf{Q} :

$$\mathbf{Q} = \mathbf{G} \cdot \text{diag}(\mathbf{K}) + \mathbf{I}$$

Hence beam offset becomes

$$\mathbf{y} = -\mathbf{Q} \cdot \mathbf{Y}$$

\mathbf{G} is lower diagonal:

$$\mathbf{G} = \begin{pmatrix} 0 & 0 & 0 & 0 & \dots \\ g_{21} & 0 & 0 & 0 & \dots \\ g_{31} & g_{32} & 0 & 0 & \dots \\ g_{41} & g_{42} & g_{43} & 0 & \dots \\ \vdots & \vdots & \vdots & \vdots & \ddots \end{pmatrix}$$

Dispersive Emittance Growth

Consider effects of finite energy spread in beam δ_{RMS}

chromatic response matrix: $\mathbf{Q}(\delta) = \mathbf{G}(\delta) \cdot \text{diag}\left(\frac{\mathbf{K}}{1+\delta}\right) + \mathbf{I}$

$$\mathbf{G}(\delta) = \mathbf{G}(0) + \left. \frac{\partial \mathbf{G}}{\partial \delta} \right|_{\delta=0} \delta$$

$$R_{34}(\delta) = R_{34}(0) + T_{346} \delta$$

↑ lattice chromaticity ↑ dispersive kicks

dispersive orbit:

$$\boldsymbol{\eta}_y \approx \frac{\Delta \mathbf{y}(\delta)}{\delta} = -[\mathbf{Q}(\delta) - \mathbf{Q}(0)] \cdot \mathbf{Y}$$

What do we measure?

BPM readings contain additional errors:

$\mathbf{b}_{\text{offset}}$ static offsets of monitors wrt quad centres

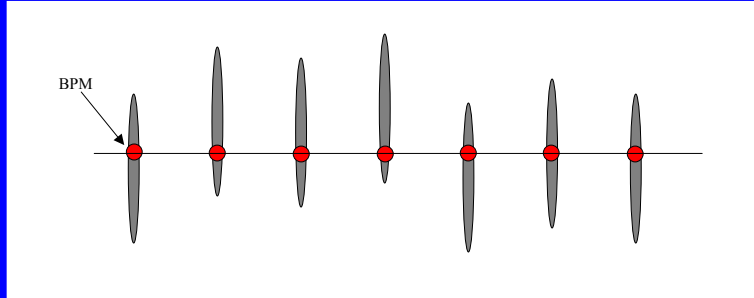
$\mathbf{b}_{\text{noise}}$ one-shot measurement noise (resolution σ_{RES})

$$\mathbf{y}_{\text{BPM}} = -\mathbf{Q} \cdot \mathbf{Y} + \mathbf{b}_{\text{offset}} + \mathbf{b}_{\text{noise}} + \mathbf{R} \cdot \mathbf{y}_0 \quad \mathbf{y}_0 = \begin{pmatrix} y_0 \\ y'_0 \end{pmatrix}$$

fixed from shot to shot random (can be averaged to zero) launch condition

In principle: all BBA algorithms deal with $\mathbf{b}_{\text{offset}}$

Scenario 1: Quad offsets, but BPMs aligned



Assuming:

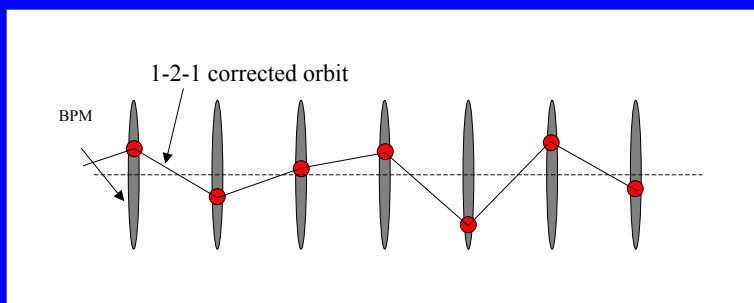
- a BPM adjacent to each quad

- a 'steerer' at each quad

simply apply one to one steering to orbit

steerer {
quad mover
dipole corrector

Scenario 2: Quads aligned, BPMs offset



one-to-one correction BAD!

Resulting orbit not Dispersion Free \Rightarrow emittance growth

Need to find a steering algorithm which effectively puts
BPMs on (some) reference line

real world scenario: some mix of scenarios 1 and 2

BBA

- **Dispersion Free Steering (DFS)**
 - Find a set of steerer settings which minimise the dispersive orbit
 - in practise, find solution that minimises difference orbit when ‘energy’ is changed
 - Energy change:
 - true energy change (adjust linac phase)
 - scale quadrupole strengths
- **Ballistic Alignment**
 - Turn off accelerator components in a given section, and use ‘ballistic beam’ to define reference line
 - measured BPM orbit immediately gives $\mathbf{b}_{\text{offset}}$ wrt to this line

DFS

Problem:

$$\Delta \mathbf{y} = - \left[\mathbf{Q} \left(\frac{\Delta E}{E} \right) - \mathbf{Q}(0) \right] \left(\frac{\Delta E}{E} \right) \cdot \mathbf{Y}$$
$$\equiv \mathbf{M} \left(\frac{\Delta E}{E} \right) \cdot \mathbf{Y}$$

Note: taking difference orbit $\Delta \mathbf{y}$ removes $\mathbf{b}_{\text{offset}}$

Solution (trivial):

$$\mathbf{Y} = \mathbf{M}^{-1} \cdot \Delta \mathbf{y}$$

Unfortunately, not that easy because of noise sources:

$$\Delta \mathbf{y} = \mathbf{M} \cdot \mathbf{Y} + \mathbf{b}_{\text{noise}} + \mathbf{R} \cdot \mathbf{y}_0$$

DFS example

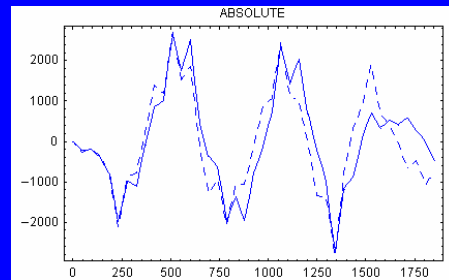
300 μm random
quadrupole errors

20% $\Delta E/E$

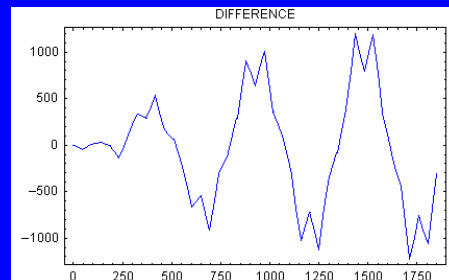
No BPM noise

No beam jitter

μm



μm

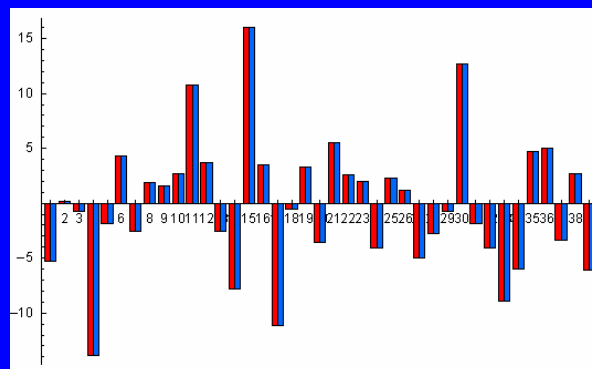


DFS example

Simple solve

$$\mathbf{Y} = \mathbf{M}^{-1} \cdot \Delta\mathbf{y}$$

In the absence of
errors, works
exactly



Resulting orbit is flat

\Rightarrow Dispersion Free

(perfect BBA)

■ original quad errors

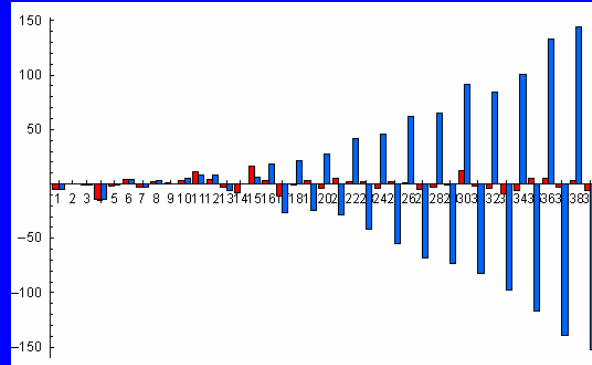
□ fitter quad errors

Now add 1 μm random BPM noise to measured difference orbit

DFS example

Simple solve

$$\mathbf{Y} = \mathbf{M}^{-1} \cdot \Delta \mathbf{y}$$



Fit is ill-conditioned!

■ original quad errors

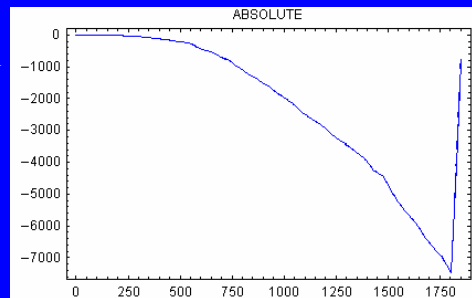
□ fitter quad errors

DFS example

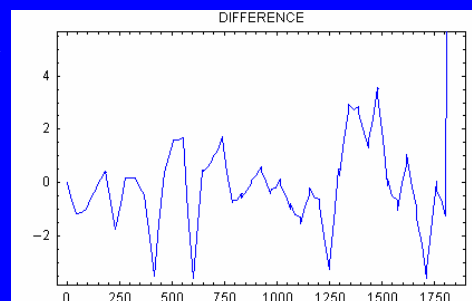
Solution is still Dispersion Free

but several mm off axis!

μm



μm



DFS: Problems

- Fit is ill-conditioned
 - with BPM noise DF orbits have very large unrealistic amplitudes.
 - Need to constrain the absolute orbit

minimise
$$\frac{\Delta \mathbf{y} \cdot \Delta \mathbf{y}^T}{2\sigma_{\text{res}}^2} + \frac{\mathbf{y} \cdot \mathbf{y}^T}{\sigma_{\text{res}}^2 + \sigma_{\text{offset}}^2}$$



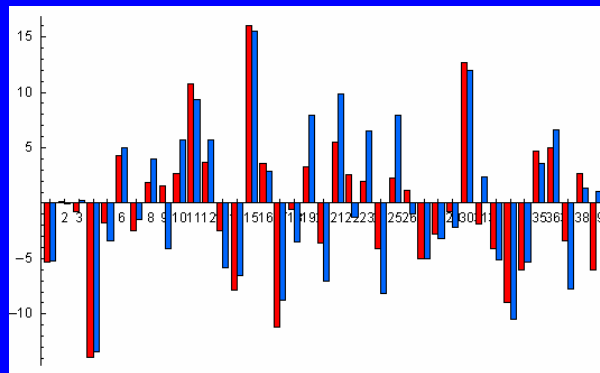
- Sensitive to initial launch conditions $\mathbf{R} \cdot \mathbf{y}_0$ (steering, beam jitter)
 - need to be fitted out or averaged away

DFS example

Minimise

$$\frac{\Delta \mathbf{y} \cdot \Delta \mathbf{y}^T}{2\sigma_{\text{res}}^2} + \frac{\mathbf{y} \cdot \mathbf{y}^T}{\sigma_{\text{res}}^2 + \sigma_{\text{offset}}^2}$$

↑
absolute orbit now constrained



remember

$$\sigma_{\text{res}} = 1\mu\text{m}$$

$$\sigma_{\text{offset}} = 300\mu\text{m}$$

■ original quad errors

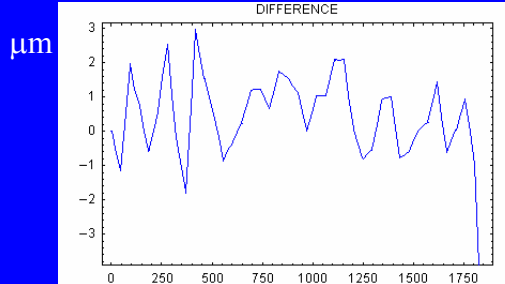
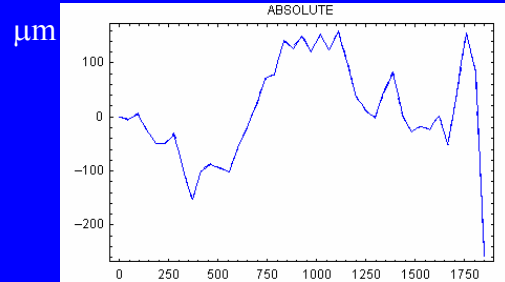
□ fitter quad errors

DFS example

Solutions much better behaved!

! Wakefields !

Orbit *not quite*
Dispersion Free, but very close



DFS practicalities

- Need to align linac in sections (bins), generally overlapping.
- Changing energy by 20%
 - quad scaling: only measures dispersive kicks from quads. Other sources ignored (not measured)
 - Changing energy upstream of section using RF better, but beware of RF steering (see initial launch)
 - dealing with energy mismatched beam may cause problems in practise (apertures)
- Initial launch conditions still a problem
 - coherent β -oscillation looks like dispersion to algorithm.
 - can be random jitter, or RF steering when energy is changed.
 - need good resolution BPMs to fit out the initial conditions.
- Sensitive to model errors (**M**)

Ballistic Alignment

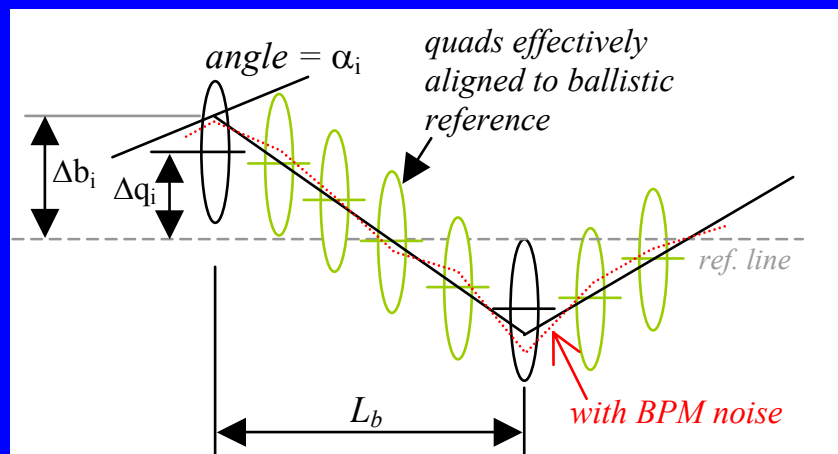
- Turn off all components in section to be aligned [magnets, and RF]
- use 'ballistic beam' to define straight reference line (BPM offsets)

$$y_{\text{BPM},i} = y_0 + s_i y'_0 + b_{\text{offset},i} + b_{\text{noise},i}$$

- Linearly adjust BPM readings to arbitrarily zero last BPM
- restore components, steer beam to adjusted ballistic line

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Ballistic Alignment



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Ballistic Alignment: Problems

- Controlling the downstream beam during the ballistic measurement
 - large beta-beat
 - large coherent oscillation
- Need to maintain energy match
 - scale downstream lattice while RF in ballistic section is off
- use feedback to keep downstream orbit under control

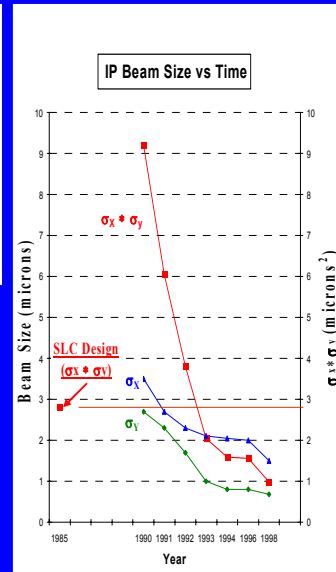
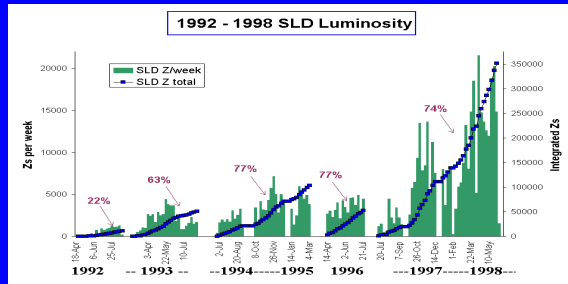
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Lecture 9: b) Lessons learnt from SLC

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USPAS, Santa Barbara, 16th-27th June, 2003

Lessons from the SLC



New Territory in Accelerator Design and Operation

- Sophisticated on-line modeling of non-linear physics.
- Correction techniques expanded from first-order (trajectory) to include second-order (emittance), and from hands-on by operators to fully automated control.
- Slow and fast feedback theory and practice.

D. Burke, SLAC

The SLC

	1980	1998	Units
Beam charge	7.2e10	4.2e10	e^\pm/bunch
Rep. rate	180	120	Hz
DR ϵ_x	3.0e-5	3.0e-5	m rad
DR ϵ_y	3.0e-5	3.0e-6	m rad
FF ϵ_x	4.2e-5	5.5e-5	m rad
FF ϵ_y	4.2e-5	1.0e-5	m rad
IP σ_x	1.65	1.4	μm
IP σ_y	1.65	0.7	μm
Pinch factor	220%	220%	Hd
Luminosity	6e30	3e30	$\text{cm}^{-2}\text{sec}^{-1}$

note: SLC was a single bunch machine ($n_b = 1$)

taken from *SLC – The End Game* by R. Assmann *et al*, proc. EPAC 2000

SLC: lessons learnt

- Control of wakefields in linac
 - orbit correction, closed (tuning) bumps
 - the need for continuous emittance measurement (automatic wire scanner profile monitors)
- Orbit and energy feedback systems
 - many MANY feedback systems implemented over the life time of the machine
 - operator ‘tweaking’ replaced by feedback loop
- Final focus optics and tuning
 - efficient algorithms for tuning (focusing) the beam size at the IP
 - removal (tuning) of optical aberrations using orthogonal knobs.
 - improvements in optics design
- many many more!

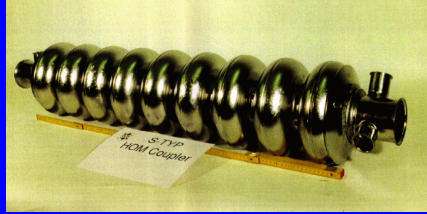
The SLC was an 10 year accelerator R&D project that also did some physics ☺

The Alternatives

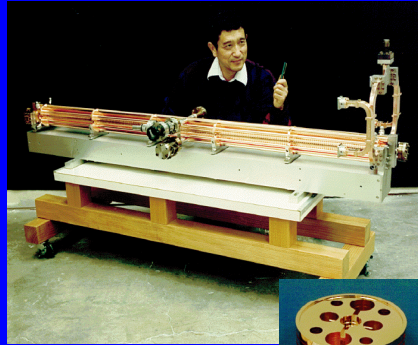
2003 $E_{cm} = 500 \text{ GeV}$

		TESLA	JLC-C	JLC-X/NLC	CLIC
f	GHz	1.3	5.7	11.4	30.0
L	$\times 10^{33} \text{ cm}^{-2}\text{s}^{-1}$	34	14	20	21
P_{beam}	MW	11.3	5.8	6.9	4.9
P_{AC}	MW	140	233	195	175
$\gamma\epsilon_y$	$\times 10^{-8}\text{m}$	3	4	4	1
σ_y^*	nm	5	4	3	1.2

Examples of LINAC technology

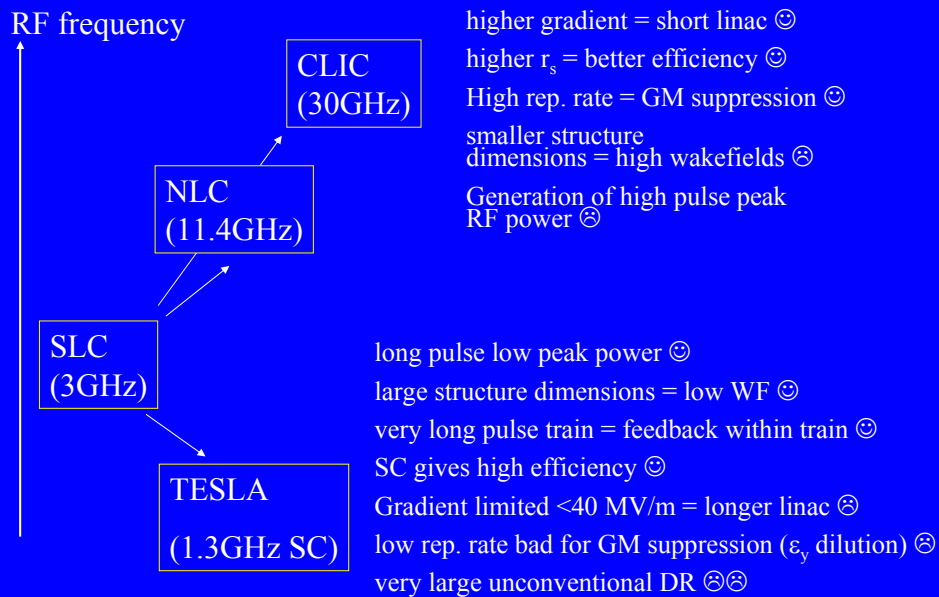


9 cell superconducting
Niobium cavity for
TESLA (1.3GHz)



11.4GHz
structure for NLCTA
(note older 1.8m structure)

Competing Technologies: swings and roundabouts



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Lecture 9: c) Summary

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The *Luminosity* Issue

$$L \propto \frac{\eta_{RF} P_{RF}}{E_{cm}} \sqrt{\frac{\delta_{BS}}{\epsilon_{n,y}}} H_D \quad \beta_y \approx \sigma_z$$

- high RF-beam conversion efficiency η_{RF}
- high RF power P_{RF}
- small normalised vertical emittance $\epsilon_{n,y}$
- strong focusing at IP (small β_y and hence small σ_z)
- could also allow higher beamstrahlung δ_{BS} if willing to live with the consequences
- Valid for low beamstrahlung regime ($\Upsilon < 1$)

High Beamstrahlung Regime

low beamstrahlung regime $\Upsilon \ll 1$:

$$L \propto P_{\text{beam}} \frac{\sqrt{\delta_{\text{BS}}}}{\sqrt{\epsilon_{y,n}}}$$

high beamstrahlung regime $\Upsilon \gg 1$:

$$L \propto P_{\text{beam}} \frac{\delta_{\text{BS}}^{3/2}}{\sigma_z \sqrt{\epsilon_{y,n}}}$$

with

$$\beta_y^* \approx \sigma_z$$

Pinch Enhancement

$$L \propto \frac{\eta_{\text{RF}} P_{\text{RF}}}{E_{\text{cm}}} \sqrt{\frac{\delta_{\text{BS}}}{\epsilon_{n,y}}} H_D$$

$$H_D = H_D(D_y)$$

$$D_y = \frac{\sigma_z}{f_{\text{beam}}} \approx \frac{2N_b r_e \sigma_z}{\gamma \sigma_y (\sigma_x + \sigma_y)} \approx \frac{2N_b r_e \sigma_z}{\gamma \sigma_y \sigma_x}$$

Trying to push hard on D_y to achieve larger H_D leads to single-bunch kink instability \rightarrow detrimental to luminosity

The Linear Accelerator (LINAC)

- Gradient given by *shunt impedance*:

- P_{RF} RF power /unit length
- r_l shunt impedance /unit length

$$E_z(z) = \sqrt{P_{RF}(z)r_l}$$

- The cavity Q defines the *fill time*:

- $v_g = \text{group velocity}$,
- $l_s = \text{structure length}$

$$t_{fill} = \begin{cases} 2Q/\omega & \text{SW} \\ 2\tau Q/\omega = l_s/v_g & \text{TW} \end{cases}$$

- For TW, τ is the structure attenuation constant:

$$P_{RF,out} = P_{RF,in} e^{-2\tau}$$

- RF power lost along structure (TW):

$$\frac{dP_{RF}}{dz} = -\frac{E_z^2}{r_l} - i_b E_z$$

power lost to structure

beam loading

η_{RF}

would like R_s to be as high as possible

$$R_s \propto \sqrt{\omega}$$

The Linear Accelerator (LINAC)

For constant gradient structures:

$$V_u = \sqrt{r_l P_0 L (1 - e^{-2\tau})}$$

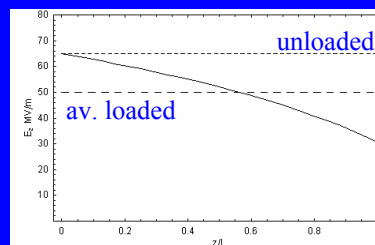
unloaded structure voltage

$$V_l = V_u - \frac{1}{2} r_l L i_{beam} \left(1 - \frac{2\tau e^{-2\tau}}{1 - e^{-2\tau}} \right)$$

loaded structure voltage (steady state)

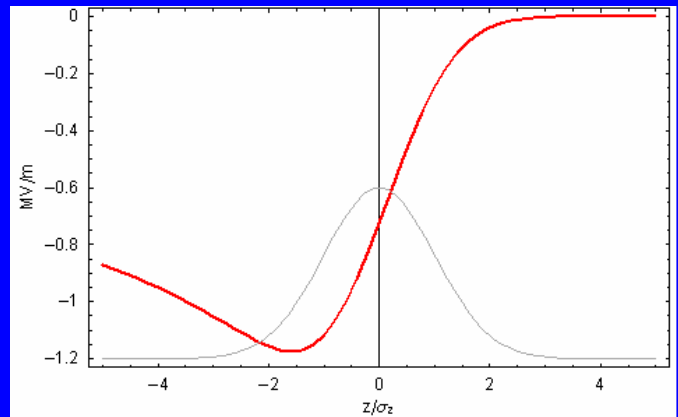
$$i_{opt} = \sqrt{\frac{P_0}{r_l L} \frac{(1 - e^{-2\tau})^{3/2}}{1 - (1 + 2\tau)e^{-2\tau}}}$$

optimum current (100% loading)



The Linear Accelerator (LINAC)

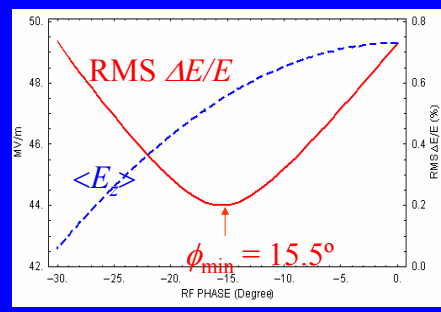
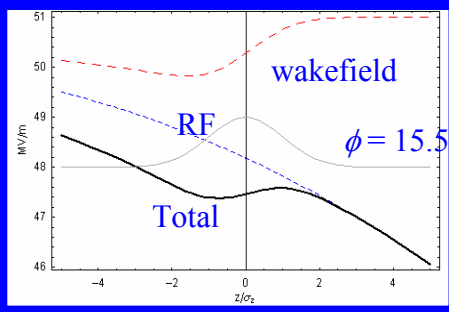
Single bunch beam loading: the Longitudinal wakefield



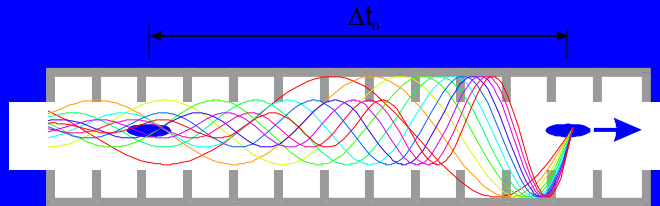
NLC X-band structure: $\langle \Delta E_z \rangle_{bunch} \approx 700 \text{ kV/m}$

The Linear Accelerator (LINAC)

Single bunch beam loading Compensation using RF phase



Transverse Wakes: The Emittance Killer!



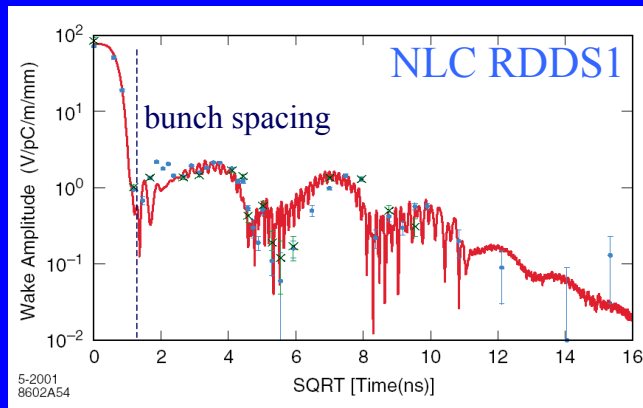
$$V(\omega, t) = I(\omega, t)Z(\omega, t)$$

Bunch current also generates transverse deflecting modes when bunches are not on cavity axis

Fields build up resonantly: latter bunches are kicked transversely

⇒ multi- and single-bunch beam breakup (MBBU, SBBU)

Damped & Detuned Structures



$$\Delta t \approx \frac{2Q_{\text{HOM}}}{\Delta\omega}$$

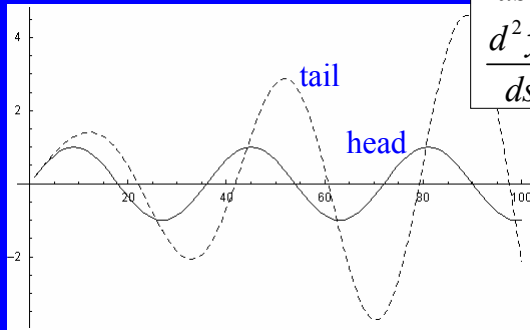
Slight random detuning between cells causes HOMs to decohere.

Will re-cohere later: needs to be damped (HOM dampers)

Single bunch wakefields

Effect of coherent betatron oscillation

- head *resonantly* drives the tail



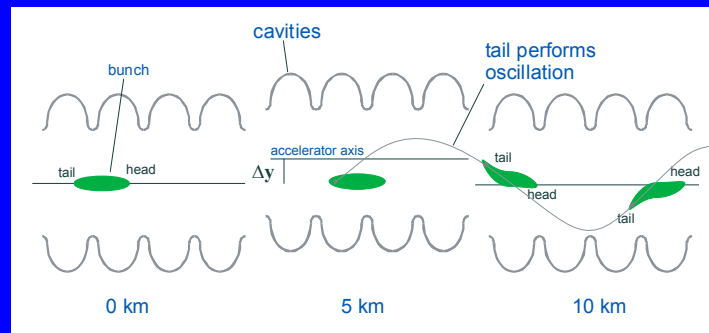
$$\frac{d^2 y_h}{ds} + k^2 y_h = 0 \quad \text{head}$$

$$\frac{d^2 y_t}{ds} + k^2 y_t = -w_{wf} y_h \quad \text{tail}$$

Cancel using BNS damping:

$$\delta_{\text{BNS}} \approx \frac{1}{16} \frac{W_{\perp}' q \sigma_z}{E} \frac{L_{\text{cell}}^2}{\sin^2(\pi \nu_{\beta})}$$

Wakefields (alignment tolerances)



$$\delta Y_{\text{RMS}} \propto \frac{1}{NW_{\perp}} \sqrt{\frac{E_z}{\beta}}$$

$$\propto \frac{f^{-3}}{N} \sqrt{\frac{E_z}{\beta}}$$

higher frequency = stronger wakefields

-higher gradients

-stronger focusing (smaller β)

-smaller bunch charge

Damping Rings

initial emittance
(~0.01m for e⁺)

$$\epsilon_f = \epsilon_{eq} + (\epsilon_i - \epsilon_{eq})e^{-2T/\tau_D}$$

final emittance equilibrium emittance damping time

$$\Delta E_{arc} = C_\gamma \frac{E^4}{\rho}$$

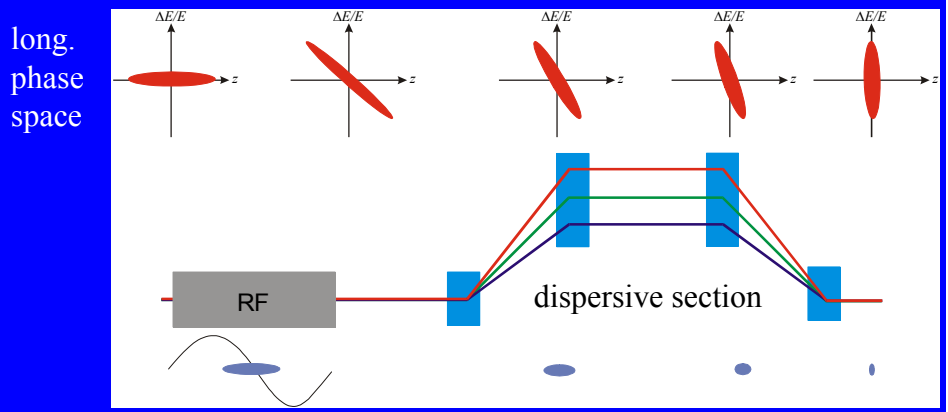
$$\tau_D = \frac{2E}{P_\gamma} \leq \frac{n_{train}}{8f_{rep}}$$

$$C = n_{train} n_b \Delta t_b c$$

$$\Delta E_{wig} \approx 1.27 \times 10^{-6} \langle B^2 \rangle (T) E^2 (\text{GeV}) L_{wig} (\text{m})$$

Bunch Compression

- bunch length from ring ~ few mm
- required at IP 100-300 μm



The linear bunch compressor

see lecture 6

initial (uncorrelated) momentum spread:	δ_u
initial bunch length	$\sigma_{z,0}$
compression ration	$F_c = \sigma_{z,0} / \sigma_z$
beam energy	E
RF induced (correlated) momentum spread:	δ_c
RF voltage	V_{RF}
RF wavelength	$\lambda_{RF} = 2\pi / k_{RF}$
longitudinal dispersion:	R_{56}

conservation of longitudinal emittance

$$F_c = \frac{\sqrt{\delta_c^2 + \delta_u^2}}{\delta_u} \Leftrightarrow \delta_c = \delta_u \sqrt{F_c^2 - 1}$$

RF cavity

$$\delta_c \approx \frac{k_{RF} V_{RF} \sigma_{z,0}}{E} \Leftrightarrow V_{RF} = \frac{E \delta_c}{k_{RF} \sigma_{z,0}} = \frac{E}{k_{RF}} \left(\frac{\delta_u}{\sigma_{z,0}} \right) \sqrt{F_c^2 - 1}$$

The linear bunch compressor

chicane (dispersive section)

$$R_{56} = -\frac{\langle \delta z \rangle}{\delta^2} = -\frac{\delta_c \sigma_{z,0}}{F^2 \delta_u^2} = \frac{k_{RF} V_{RF}}{E} \left(\frac{\sigma_{z,0}}{\delta_u} \right)^2 \frac{1}{F^2}$$

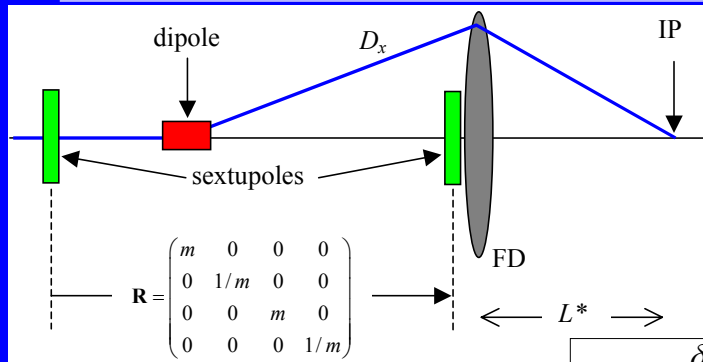
$$R_{56} \begin{cases} < 0 & \text{wiggler} \\ > 0 & \text{arc} \end{cases}$$

Two stage compression used to reduce final energy spread



$$\begin{aligned} \delta_f &= (F_1 \delta_i) \cdot \left(\frac{E_0}{E_0 + \Delta E} \right) \cdot F_2 \\ &= F_T \delta_i \left(\frac{E_0}{E_0 + \Delta E} \right) \end{aligned}$$

Final Focusing



chromatic correction $S = \frac{1}{L^* \eta}$

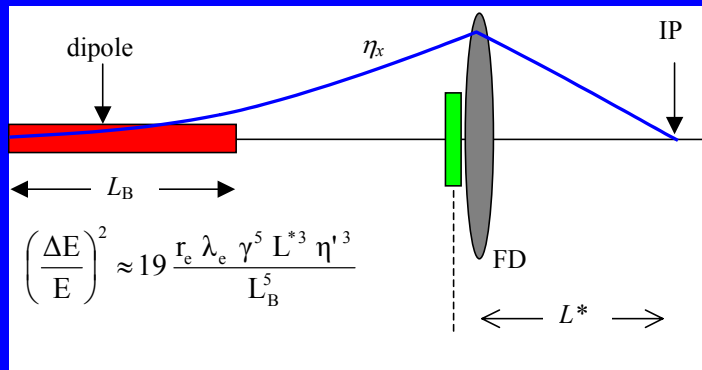
$$\Delta y'_{FD} = -\frac{\delta y}{L^*}$$

$$\Delta x'_{FD} = -\frac{\delta(x + \eta\delta)}{L^*}$$

$$\Delta y'_{SX} = S(x + \eta\delta)y$$

$$\Delta x'_{SX} = -S[(x + \eta\delta)^2 - y^2]$$

Synchrotron Radiation effects



$$\frac{\Delta \sigma_y^{*2}}{\sigma_y^{*2}} \approx W_y^2 \left(\frac{\Delta E}{E}\right)^2$$

FD chromaticity + dipole SR sets limits on minimum bend length

Final Focusing: Fundamental limits

Already mentioned that $\beta_y \geq \sigma_z$

At high-energies, additional limits set by so-called *Oide Effect*: synchrotron radiation in the final focusing quadrupoles leads to a beamspace growth at the IP

minimum beam size: $\sigma \approx 1.83 (r_e \hat{\lambda}_e F)^{1/4} \varepsilon_n^{3/4}$ *independent of E!*

occurs when $\beta \approx 2.39 (r_e \hat{\lambda}_e F)^{3/4} \gamma \varepsilon_n^{3/4}$

F is a function of the focusing optics: typically $F \sim 7$ (minimum value ~ 0.1)

LINAC quadrupole stability

$$y^* = \sum_{i=1}^{N_Q} k_{Q,i} \Delta Y_i g_i = k_Q \sum_{i=1}^{N_Q} \Delta Y_i g_i$$

$$g_i = \sqrt{\frac{\gamma_i}{\gamma^*}} \sqrt{\beta_i \beta^*} \sin(\Delta \phi_i)$$

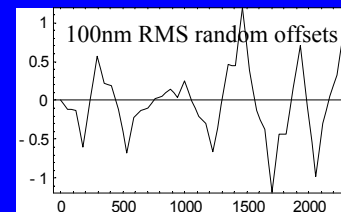
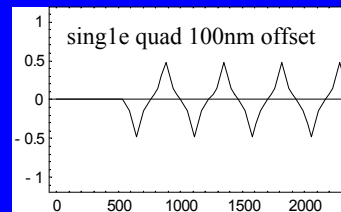
for uncorrelated offsets

$$\langle y^{*2} \rangle = \frac{\beta^* \langle \Delta Y^2 \rangle}{\gamma^*} \sum_{i=1}^{N_Q} \gamma_i k_{Q,i}^2 \beta_i \sin^2(\Delta \phi_i)$$

Dividing by $\sigma_y^{*2} = \beta^* \varepsilon_{y,n} / \gamma^*$ and taking average values:

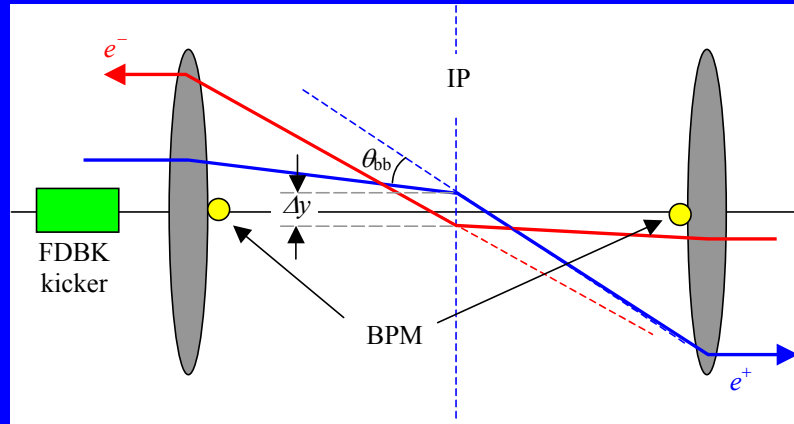
$$\frac{\langle y_j^2 \rangle}{\sigma_y^{*2}} \approx \frac{N_Q k_Q^2 \bar{\beta} \bar{\gamma}}{2 \varepsilon_{y,n}} \sigma_{\Delta Y}^2 \leq 0.3^2$$

take $N_Q = 400$, $\varepsilon_y \sim 6 \times 10^{-14}$ m, $\beta \sim 100$ m, $k_1 \sim 0.03$ m⁻¹ $\Rightarrow \sim 25$ nm



Beam-Beam orbit feedback

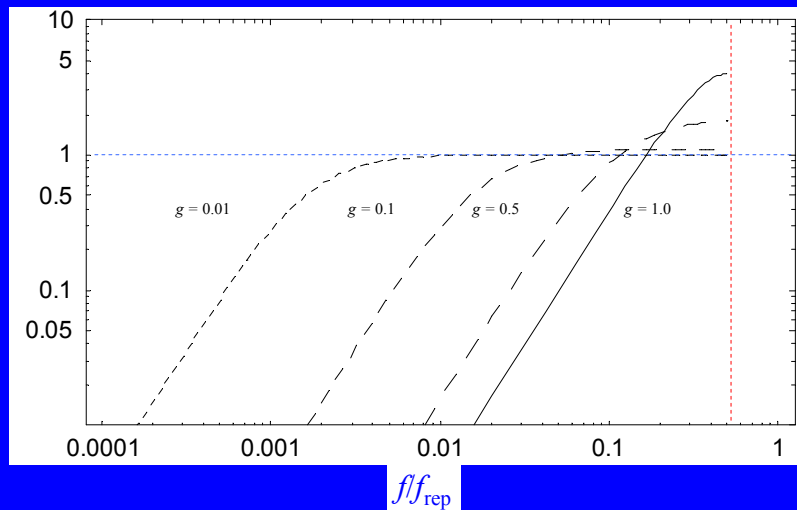
see lecture 8



use strong beam-beam kick to keep beams colliding

Generally, orbit control (feedback) will be used extensively in LC

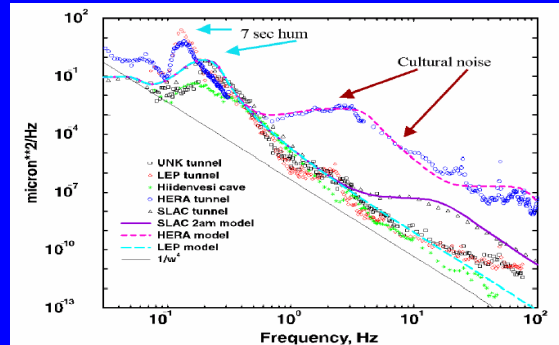
Beam based feedback: bandwidth



Good rule of thumb: attenuate noise with $f < f_{rep}/20$

Ground motion spectra

Both frequency spectrum and *spatial* correlation important for LC performance



2D power spectrum

measurable relative power spectrum

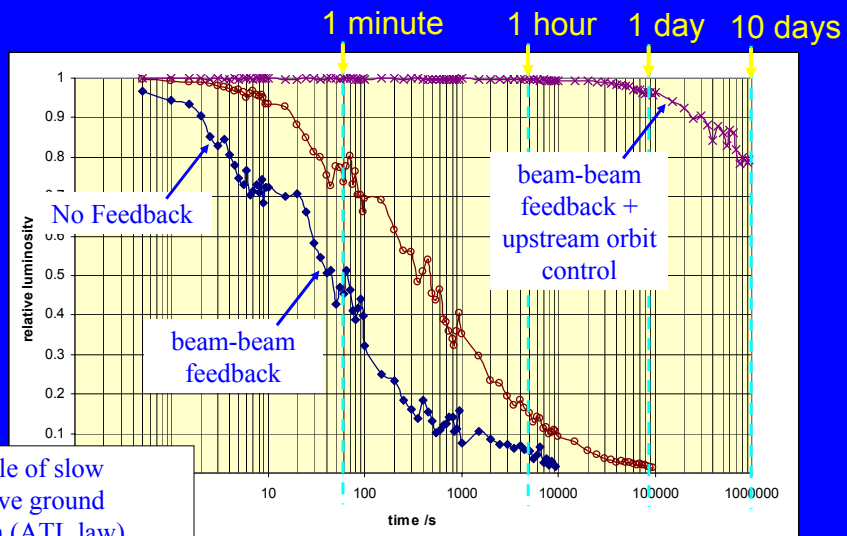
$$P(\omega, k)$$

$$\rho(\omega, L) = \frac{1}{\pi} \int_{-\infty}^{+\infty} P(\omega, k) [1 - \cos(kL)] dk$$

Long Term Stability

see lecture 8

understanding of ground motion and vibration spectrum important



example of slow diffusive ground motion (ATL law)

A Final Word

- Technology decision due 2004
- Start of construction 2007+
- First physics 2012++
- There is *still* much to do!

WE NEED YOUR HELP

for

the Next Big Thing

hope you enjoyed the course

Nick, Andy, Andrei and PT